

**Only one option correct**

- Two bodies of mass  $m$  and  $M$  are initially at rest at very large distance from each other. They approach each other under the influence of mutual gravitational force. Their relative velocity of approach at a distance of separation  $d$  is
  - $\sqrt{\frac{2G(M+m)}{d^2}}$
  - $\sqrt{\frac{G(M+m)}{d^2}}$
  - $\sqrt{\frac{2G(M+m)}{d}}$
  - $\sqrt{\frac{G(M+m)}{d}}$
- Two planets A and B have the same material density. If the radius of A is twice that of B, then the ratio of escape velocities from the planets  $v_A : v_B$  is
  - 2
  - $\sqrt{2}$
  - $1/\sqrt{2}$
  - $1/2$
- A body is projection horizontally from the surface of the Earth of radius  $R$  with a velocity equal to  $n$  times the escape velocity. Neglect rotational effects of the earth. If the maximum height attained by the body from the Earth's surface is  $R/2$  then,  $n$  is equal to
  - $\sqrt{0.6}$
  - $\sqrt{3/2}$
  - $\sqrt{0.4}$
  - $1/\sqrt{2}$
- A particle is projected from the mid-point of the line joining two fixed particles each of mass  $m$ . If the distance of separation between the fixed particles is  $l$ , the minimum velocity of projection of the particle so as to escape is equal to
  - $\sqrt{\frac{GM}{l}}$
  - $\sqrt{\frac{GM}{2l}}$
  - $\sqrt{\frac{2GM}{l}}$
  - $2\sqrt{\frac{2GM}{l}}$
- A satellite is launched into a circular orbit of radius  $R$  around the earth while a second satellite is launched into an orbit of radius  $1.02 R$ . the percentage difference in the time periods is nearly
  - 0.7
  - 1.0
  - 1.5
  - 3
- A particle is released from infinity. Another particle is released from a height of  $R/2$  ( where  $R$  is the radius of the earth ). The ratio of their velocities as they reach the surface of the earth is
  - $\sqrt{3} : 1$
  - 3:2
  - 1:1
  - 1:2
- A ball A of mass  $m$  falls to the surface of the earth from infinity. Another ball B of mass  $2m$  falls to the earth from the height equal to six times the radius of the earth. Ratio of velocity of A to that of B on reaching the earth is
  - $\sqrt{\frac{6}{5}}$
  - $\sqrt{\frac{5}{4}}$
  - $\sqrt{\frac{5}{7}}$
  - $\sqrt{\frac{7}{6}}$
- A satellite is revolving around the earth with a speed of  $v_0$ . If it stops suddenly, the velocity with which it reaches the surface of the earth is (  $v_e$  is the escape velocity from the surface of the earth )
  - $\frac{v_e^2}{v_0}$
  - $v_0$
  - $\sqrt{v_e^2 - v_0^2}$
  - $\sqrt{v_e^2 - 2v_0^2}$
- An artificial satellite is moving around the earth in a circular orbit with a speed equal to half the magnitude of escape velocity. The height of the satellite from the surface of the earth is
  - $2R$
  - $R/2$
  - $R$
  - $R/4$
- Three identical bodies, each of mass  $M$ , are moving in a circle of radius  $R$  under the influence of mutual gravitational force. The speed of each body is
  - $\sqrt{\frac{GM}{R}}$
  - $\sqrt{\frac{GM}{3R}}$
  - $\sqrt{\frac{GM}{R\sqrt{3}}}$
  - $\sqrt{\frac{GM}{9R}}$
- Four particles of equal masses  $M$  move along a circle of radius  $R$  under the action of their mutual gravitational attraction maintaining a square shape. The speed of each particle is
  - $\sqrt{\frac{GM}{R} \left( \frac{2\sqrt{2}+1}{4} \right)}$
  - $\sqrt{\frac{GM}{R} \left( \frac{\sqrt{2}+1}{4} \right)}$
  - $\sqrt{\frac{GM}{R} \left( \frac{1}{\sqrt{2}+1} \right)}$
  - $\sqrt{\frac{4GM}{R} \left( \frac{1}{\sqrt{2}+1} \right)}$
- Three point masses, each of mass  $M$  each, are moving in a circle under their mutual gravitational attractive forces. If speed of each body is  $v$  then the distance between any two bodies is
  - $\frac{2GM}{v^2}$
  - $\frac{3GM}{v^2}$
  - $\frac{\sqrt{3}GM}{v^2}$
  - $\frac{GM}{v^2}$
- Two point masses of mass  $4m$  and  $m$  respectively separated by  $d$  distance are revolving under mutual force of attraction. Ratio of their kinetic energies will be
  - 1:4
  - 1:5
  - 1:1
  - 1:2
- If the kinetic energy of a satellite ( in a stationary orbit ) is  $2M$  J, its total energy is
  - 1M J
  - 2M J
  - 4M J
  - 8M J

15. A space ship of mass  $m$  is in circular orbit of radius  $2R_e$  about the earth of mass  $M$  and radius  $R_e$ . Energy required to transfer the space ship to circular orbit of radius  $3R_e$  is
1.  $\frac{GmM}{8R_e}$
  2.  $\frac{GmM}{4R_e}$
  3.  $\frac{GmM}{24R_e}$
  4.  $\frac{GmM}{12R_e}$
16. A satellite of mass  $m$  is moving in a circular orbit of radius  $r$  around a planet of mass  $M$ . Its angular momentum is
1.  $\sqrt{Gm^2Mr}$
  2.  $\sqrt{GmM^2r}$
  3.  $\sqrt{\frac{GmM^2}{r}}$
  4.  $\sqrt{Gm^2Mr^2}$
17. Kinetic energy of a satellite in its orbit around the earth is  $E$ . Additional kinetic energy required by the satellite to escape the earth's orbit is
1.  $4E$
  2.  $2E$
  3.  $\sqrt{2}E$
  4.  $E$
18. A satellite of mass  $m$ , initially at rest on the earth, is launched into a circular orbit at a height equal to the radius of the earth. The minimum energy required is
1.  $\frac{\sqrt{3}}{4} mgR$
  2.  $\frac{1}{2} mgR$
  3.  $\frac{1}{4} mgR$
  4.  $\frac{3}{4} mgR$
19. Assertion (A) : There is no atmosphere on moon  
Reason (R) : RMS velocity of molecules of gases is less than the escape velocity on surface of the moon
1. A is correct and R is the correct explanation for it
  2. A is correct but R is not the correct explanation for A
  3. A is wrong but R is generally correct
  4. Both A and R are wrong
20. The maximum energy required to launch a satellite of mass  $m$  ( from the earth's surface ) into a circular orbit of radius  $3R$  is ( $R$  is the radius of the earth )
1.  $\frac{5}{6} mgR$
  2.  $\frac{3}{8} mgR$
  3.  $\frac{3}{4} mgR$
  4.  $\frac{7}{8} mgR$

**Key**

- 1. 3
- 2. 2
- 3. 1
- 4. 4
- 5. 4
- 6. 1
- 7. 4
- 8. 4
- 9. 3
- 10. 3
- 11. 1
- 12. 4
- 13. 1
- 14. 2
- 15. 4
- 16. 1
- 17. 4
- 18. 4
- 19. 2
- 20. 1