

Only one option correct

1. Two bodies of mass m and M are initially at rest at very large distance from each other. They approach each other under the influence of mutual gravitational force. Their relative velocity of approach at a distance of separation d is

1. $\sqrt{\frac{2G(M+m)}{d^2}}$
2. $\sqrt{\frac{G(M+m)}{d^2}}$
3. $\sqrt{\frac{2G(M+m)}{d}}$
4. $\sqrt{\frac{G(M+m)}{d}}$

2. Two planets A and B have the same material density. If the radius of A is twice that of B, then the ratio of escape velocities from the planets $v_A : v_B$ is

1. 2
2. $\sqrt{2}$
3. $1/\sqrt{2}$
4. 1/2

3. A body is projected horizontally from the surface of the Earth of radius R with a velocity equal to n times the escape velocity. Neglect rotational effects of the earth. If the maximum height attained by the body from the Earth's surface is $R/2$ then, n is equal to

1. $\sqrt{0.6}$
2. $\sqrt{3}/2$
3. $\sqrt{0.4}$
4. $1/\sqrt{2}$

4. A particle is projected from the mid-point of the line joining two fixed particles each of mass m . If the distance of separation between the fixed particles is l , the minimum velocity of projection of the particle so as to escape is equal to

1. $\sqrt{\frac{GM}{l}}$
2. $\sqrt{\frac{GM}{2l}}$
3. $\sqrt{\frac{2GM}{l}}$
4. $2\sqrt{\frac{2GM}{l}}$

5. A satellite is launched into a circular orbit of radius R around the earth while a second satellite is launched into an orbit of radius $1.02 R$. the percentage difference in the time periods is nearly

1. 0.7
2. 1.0
3. 1.5
4. 3

6. A particle is released from infinity. Another particle is released from a height of $R/2$ (where R is the radius of the earth). The ratio of their velocities as they reach the surface of the earth is

1. $\sqrt{3} : 1$
2. 3:2
3. 1:1
4. 1:2

7. A ball A of mass m falls to the surface of the earth from infinity. Another ball B of mass $2m$ falls to the earth from the height equal to six times the radius of the earth. Ratio of velocity of A to that of B on reaching the earth is

1. $\sqrt{\frac{6}{5}}$
2. $\sqrt{\frac{5}{4}}$
3. $\sqrt{\frac{5}{7}}$
4. $\sqrt{\frac{7}{6}}$

8. A satellite is revolving around the earth with a speed of v_o . If it stops suddenly, the velocity with which it reaches the surface of the earth is (v_e is the escape velocity from the surface of the earth)

1. $\frac{v_e^2}{v_o}$
2. v_o
3. $\sqrt{v_e^2 - v_o^2}$
4. $\sqrt{v_e^2 - 2v_o^2}$

9. An artificial satellite is moving around the earth in a circular orbit with a speed equal to half the magnitude of escape velocity. The height of the satellite from the surface of the earth is

1. $2R$
2. $R/2$
3. R
4. $R/4$

10. Three identical bodies, each of mass M , are moving in a circle of radius R under the influence of mutual gravitational force. The speed of each body is

1. $\sqrt{\frac{GM}{R}}$
2. $\sqrt{\frac{GM}{3R}}$
3. $\sqrt{\frac{GM}{R\sqrt{3}}}$
4. $\sqrt{\frac{GM}{9R}}$

11. Four particles of equal masses M move along a circle of radius R under the action of their mutual gravitational attraction maintaining a square shape. The speed of each particle is

1. $\sqrt{\frac{GM}{R}} \left(\frac{2\sqrt{2}+1}{4} \right)$
2. $\sqrt{\frac{GM}{R}} \left(\frac{\sqrt{2}+1}{4} \right)$
3. $\sqrt{\frac{GM}{R}} \left(\frac{1}{\sqrt{2}+1} \right)$
4. $\sqrt{\frac{4GM}{R}} \left(\frac{1}{\sqrt{2}+1} \right)$

12. Three point masses, each of mass M each, are moving in a circle under their mutual gravitational attractive forces. If speed of each body is v then the distance between any two bodies is

1. $\frac{2GM}{v^2}$
2. $\frac{3GM}{v^2}$
3. $\frac{\sqrt{3}GM}{v^2}$
4. $\frac{GM}{v^2}$

13. Two point masses of mass $4m$ and m respectively separated by d distance are revolving under mutual force of attraction. Ratio of their kinetic energies will be

1. 1:4
2. 1:5
3. 1:1
4. 1:2

14. If the kinetic energy of a satellite (in a stationary orbit) is $2M$ J, its total energy is

1. -1M J
2. -2M J
3. -4M J
4. -8M J

15. A space ship of mass m is in circular orbit of radius $2R_e$ about the earth of mass M and radius R_e . Energy required to transfer the space ship to circular orbit of radius $3R_e$ is

1. $\frac{GmM}{8R_e}$
2. $\frac{GmM}{4R_e}$
3. $\frac{GmM}{24R_e}$
4. $\frac{GmM}{12R_e}$

16. A satellite of mass m is moving in a circular orbit of radius r around a planet of mass M . Its angular momentum is

1. $\sqrt{Gm^2Mr}$
2. $\sqrt{GmM^2r}$
3. $\sqrt{\frac{GmM^2}{r}}$
4. $\sqrt{Gm^2Mr^2}$

17. Kinetic energy of a satellite in its orbit around the earth is E . Additional kinetic energy required by the satellite to escape the earth's orbit is

1. $4E$
2. $2E$
3. $\sqrt{2}E$
4. E

18. A satellite of mass m , initially at rest on the earth, is launched into a circular orbit at a height equal to the radius of the earth. The minimum energy required is

1. $\frac{\sqrt{3}}{4} mgR$
2. $\frac{1}{2} mgR$
3. $\frac{1}{4} mgR$
4. $\frac{3}{4} mgR$

19. Assertion (A) : There is no atmosphere on moon
 Reason (R) : RMS velocity of molecules of gases is less than the escape velocity on surface of the moon

1. A is correct and R is the correct explanation for it
2. A is correct but R is not the correct explanation for A
3. A is wrong but R is generally correct
4. Both A and R are wrong

20. The maximum energy required to launch a satellite of mass m (from the earth's surface) into a circular orbit of radius $3R$ is (R is the radius of the earth)

1. $\frac{5}{6} mgR$
2. $\frac{3}{8} mgR$
3. $\frac{3}{4} mgR$
4. $\frac{7}{8} mgR$

Key

1. 3
2. 2
3. 1
4. 4
5. 4
6. 1
7. 4
8. 4
9. 3
10. 3
11. 1
12. 4
13. 1
14. 2
15. 4
16. 1
17. 4
18. 4
19. 2
20. 1